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(71) Applicant (for all designated States except US): SHELL INTERNATIONALE RESEARCH MAATSCHAPPIJ B.V. [NL/NL]; Carel van Bylandlaan 30, NL-2596 HR The Hague (NL).		
(72) Inventors; and		Published
(75) Inventors/Applicants (for US only): BETTING, Marco [NL/NL]; Czaar Peterstraat 229, NL-1018 PL Amsterdam (NL). TJEENK WILLINK, Cornelis, Antonie [NL/NL]; Volmerlaan 8, NL-2288 GD Rijswijk (NL). VAN VEEN, Johannes, Miquel, Henri, Maria [NL/NL]; Czaar Peterstraat 229, NL-1018 PL Amsterdam (NL).		With international search report.
(54) Title: REMOVING A GASEOUS COMPONENT FROM A FLUID		
(57) Abstract		
<p>A method of removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components is provided. The stream is induced to flow at supersonic velocity through a conduit so as to decrease the temperature of the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, the conduit being provided with swirl imparting means to impart a swirling motion to the stream of fluid thereby inducing the particles to flow to a radially outer section of a collecting zone in the stream. A shock wave is created in the stream so as to decrease the axial velocity of the fluid to subsonic velocity, and the particles are extracted into an outlet stream from said radially outer section of the collecting zone, wherein the shock wave is created downstream the swirl imparting means and upstream the collecting zone.</p>		

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## REMOVING A GASEOUS COMPONENT FROM A FLUID

The present invention relates to removing of one or more gaseous components from a stream of fluid. More particularly the invention relates to removing of a gaseous component from a fluid stream by decreasing the 5 temperature of the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, and separating the particles from the stream. Such separation can find application in 10 various industrial processes, for example in removing of carbon dioxide from flue gas, in air-conditioning (water removal) and in drying of natural gas before distributing the gas into a network of pipelines. The term "natural gas" is applied to gas produced from underground 15 accumulations of widely varying composition. Apart from hydrocarbons, natural gas generally includes nitrogen, carbon dioxide and sometimes a small proportion of hydrogen sulphide. The principal hydrocarbon is methane, the lightest and lowest boiling member of the paraffin 20 series of hydrocarbons. Other constituents are ethane, propane, butane, pentane, hexane, heptane, etc. The lighter constituents, e.g. up to butane, are in gaseous phase at atmospheric temperatures and pressures. The heavier constituents are in gaseous phase when at 25 elevated temperatures during production from the subsurface and in liquid phase when the gas mixture has cooled down. Natural gas containing such heavier constituents is known as "wet gas" as distinct from dry gas containing none or only a small proportion of liquid 30 hydrocarbons.

Dutch patent application No. 8901841 discloses a method of removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components, wherein the stream is induced to flow at a supersonic velocity through a conduit so as to decrease the temperature of the fluid in the conduit to below the condensation point of the selected component thereby forming condensed particles of the selected component. The conduit is provided with swirl imparting means to impart a swirling motion to the stream of fluid flowing at supersonic velocity. The condensed particles are extracted in a first outlet stream from a radially outer section of the stream and the remaining fluid is collected in a second outlet stream from a central part of the stream. The velocity in said radially outer section and in said central part of the stream is supersonic.

In an embodiment of the device for separating a gas from a gas mixture as disclosed in NL-8901841, separate shock waves occur in the first and second outlet streams, leading to a relatively large flow resistance of the fluid. Furthermore, the separation efficiency is relatively low so that substantial amounts of the condensed particles are still present in the second outlet stream.

It is an object of the invention to provide an improved method and device for removing a gaseous component from a stream of fluid containing a plurality of gaseous components.

In accordance with the invention there is provided a method of removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components, the method comprising the steps of:

- inducing said stream to flow at supersonic velocity through a conduit so as to decrease the temperature of

the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, the conduit being provided with swirl  
5 imparting means to impart a swirling motion to the stream of fluid thereby inducing the particles to flow to a radially outer section of a collecting zone in the stream;

- creating a shock wave in the stream so as to decrease
- 10 the axial velocity of the fluid to subsonic velocity;
- extracting the particles into an outlet stream from said radially outer section of the collecting zone, wherein the shock wave is created downstream the swirl imparting means and upstream the collecting zone.

15 In accordance with another aspect of the invention there is provided a device for removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components, comprising:

- means for inducing said stream to flow at supersonic velocity through a conduit so as to decrease the temperature of the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, the conduit being provided with swirl imparting means to impart a swirling motion to the stream of fluid thereby inducing the particles to flow to a radially outer section of a collecting zone in the stream;
- means for creating a shock wave in the stream so as to decrease the axial velocity of the fluid to subsonic velocity;
- 20 - means for extracting the particles into an outlet stream from said radially outer section of the collecting zone, wherein the means for creating the shock wave is

arranged to create the shock wave downstream the swirl imparting means and upstream the collecting zone.

It was found that the separation efficiency is significantly improved if collection of the particles in the collecting zone takes place after the shock wave, i.e. in subsonic flow rather than in supersonic flow. This is because the shock wave dissipates a substantial amount of kinetic energy of the stream and thereby strongly reduces the axial component of the fluid velocity while the tangential component (caused by the swirl imparting means) remains substantially unchanged. As a result the density of the particles in the radially outer section of the collecting zone is significantly higher than elsewhere in the conduit where the flow is supersonic. It is believed that this effect is caused by the strongly reduced axial fluid velocity and thereby a reduced tendency of the particles to be entrained by a central "core" of the stream where the fluid flows at a higher axial velocity than nearer the wall of the conduit. Thus, in the subsonic flow regime the centrifugal forces acting on the condensed particles are not to a great extent counter-acted by the entraining action of the central "core" of the stream, so that the particles are allowed to agglomerate in the radially outer section of the collecting zone from which they are extracted.

Preferably the shock wave is created by inducing the stream of fluid to flow through a diffuser. A suitable diffuser is a supersonic diffuser.

In an advantageous embodiment, the collecting zone is located adjacent the outlet end of the diffuser.

Further drying of the fluid stream is suitably achieved by introducing the outlet stream of collected particles into a gas/liquid separator to separate a

gaseous fraction of the outlet stream from a liquid fraction thereof.

Advantageously, the gaseous fraction of the outlet stream is mixed with the stream of fluid induced to flow at supersonic velocity through the conduit.

Suitably the means for inducing the stream to flow at supersonic velocity comprises a Laval-type inlet of the conduit, wherein the smallest cross-sectional flow area of the diffuser is larger than the smallest cross-sectional flow area of the Laval-type inlet.

In an attractive embodiment of the invention the fluid forms a natural gas produced from an earth formation, and said selected temperature is defined by the condensation point of the selected component whereby condensed particles of the selected component are formed, which particles agglomerate to a liquid outlet stream.

The selected component is, for example, one or more of ethane, propane, butane, pentane, hexane, heptane and octane.

The invention will be described in more detail by way of example with reference to the drawings in which:

Fig. 1 shows schematically a longitudinal cross-section of a first embodiment of the device according to the invention; and

Fig. 2 shows schematically a longitudinal cross-section of a second embodiment of the device according to the invention.

In Fig. 1 is shown a conduit in the form of an open-ended tubular housing 1 having a fluid inlet 3 at one end of the housing, a first outlet 5 for substantially liquid near the other end of the housing, and a second outlet 7 for substantially gas at said other end of the housing. The flow-direction in the device 1 is from the inlet 3 to the first and second outlets 5, 7. The inlet 3 is of the Laval-type, having a longitudinal cross-section of

converging - diverging shape in the flow direction so as to induce a supersonic flow velocity to a fluid stream which is to flow into the housing via said inlet 3. The housing 1 is further provided with a primary cylindrical part 9 and a diffuser 11 whereby the primary cylindrical part 9 is located between the inlet 3 and the diffuser 11. One or more (e.g. four) delta-shaped wings 15 project radially inward from the inner surface of the primary cylindrical part 9, each wing 15 being arranged at a selected angle to the flow-direction in the housing so as to impart a swirling motion to fluid flowing at supersonic velocity through the primary cylindrical part 9 of the housing 1.

The diffuser 11 has a longitudinal section of converging - diverging shape in the flow direction, defining a diffuser inlet 17 and a diffuser outlet 19. The smallest cross-sectional flow area of the diffuser is larger than the smallest cross-sectional flow area of the Laval-type inlet 3.

The housing 1 further includes a secondary cylindrical part 17 having a larger flow area than the primary cylindrical part 9 and being arranged downstream the diffuser 11 in the form of a continuation of the diffuser 11. The secondary cylindrical part 17 is provided with longitudinal outlet slits 18 for liquid, which slits 18 are arranged at a suitable distance from the diffuser outlet 19.

A liquid outlet chamber 21 encloses the secondary cylindrical part 17, and is provided with the aforementioned first outlet 5 for substantially liquid.

The secondary cylindrical part 17 debouches into the aforementioned second outlet 7 for substantially gas.

Normal operation of the device 1 is now explained for the application of drying of natural gas. This is by way of example only, and the device 1 is equally suitable for

application in other process schemes where gases other than hydrocarbon are to be treated.

A stream of natural gas including lighter and heavier gaseous components, such as methane, ethane, propane, butane, pentane, hexane etc., is introduced into the Laval-type inlet 3. As the stream flows through the inlet 3, the stream is accelerated to supersonic velocity. As a result of the strongly increasing velocity of the stream, the temperature of the stream decreases to below the condensation point of the heavier gaseous components (e.g. pentane, hexane, etc.) which thereby condense to form a plurality of liquid particles. As the stream flows along the delta-shaped wings 15 a swirling motion is imparted to the stream (schematically indicated by spiral 22) so that the liquid particles become subjected to radially outward centrifugal forces. When the stream enters the diffuser 11 a shock wave is created near the downstream outlet 19 of the diffuser 11. The shock wave dissipates a substantial amount of kinetic energy of the stream, whereby mainly the axial component of the fluid velocity is decreased. As a result of the strongly decreased axial component of the fluid velocity, the central part of the stream (or "core") flows at a reduced axial velocity so that there is a reduced tendency of the condensed particles to be entrained by the central part of the stream flowing in the secondary cylindrical part 17. The condensed particles can therefore agglomerate in a radially outer section of a collecting zone of the stream in the secondary cylindrical part 17. The agglomerated particles form a layer of liquid which is extracted from the collecting zone via the outlet slits 18, the outlet chamber 21, and the first outlet 5 for substantially liquid.

The dried natural gas is discharged through the second outlet 7 for substantially gas.

In Fig. 2 is shown a second embodiment of the device for carrying out the invention, the device having an open-ended tubular housing 23 with a Laval-type fluid inlet 25 at one end and a first outlet 27 for substantially liquid at the other end of the housing. The flow-direction for fluid in the device is indicated by arrow 30. The housing has, from the inlet 25 to the liquid outlet 27, a primary substantially cylindrical part 33, a diverging diffuser 35, a secondary cylindrical part 37 and a diverging part 39. A delta-shaped wing 41 projects radially inward in the primary cylindrical part 33, the wing 37 being arranged at a selected angle to the flow-direction in the housing so as to impart a swirling motion to fluid flowing at supersonic velocity through the housing 23. A tube-shaped second outlet 43 for substantially gas extends through the first outlet 27 coaxially into the housing, and has an inlet opening 45 at the downstream end of the secondary cylindrical part 37. The outlet 43 is internally provided with a straightener (not shown), e.g. a vane-type straightener, for transferring swirling flow of the gas into straight flow.

Normal operation of the second embodiment is substantially similar to normal operation of the first embodiment, i.e. supersonic swirling flow occurs in the primary cylindrical part 33, the shock wave occurs near the transition of the diffuser 35 to the secondary cylindrical part 37, subsonic flow occurs in the secondary cylindrical part 37, the liquid is discharged through the first outlet 27, and the dried gas is discharged through the second outlet 43 in which the swirling flow of the gas is transferred into straight flow by the straightener.

In the above detailed description, the housing, the primary cylindrical part, the diffuser and the secondary

cylindrical part have a circular cross-section. However, any other suitable cross-section of each one of these items can be selected. Also, the primary and secondary parts can alternatively have a shape other than cylindrical, for example a frusto-conical shape. Furthermore, the diffuser can have any other suitable shape, for example without a converging part (as shown in Fig. 2) especially for applications at lower supersonic fluid velocities.

Instead of each wing being arranged at a fixed angle relative to the axial direction of the housing, the wing can be arranged at an increasing angle in the direction of flow, preferably in combination with a spiralling shape of the wing.

Furthermore, each wing can be provided with a raised wing-tip (also referred to as a winglet).

Instead of the diffuser having a diverging shape (Fig. 2), the diffuser alternatively has a diverging section followed by a converging section when seen in the flow direction. An advantage of such diverging - converging shaped diffuser is that less fluid temperature increase occurs in the diffuser.

To increase the size of the condensed particles, the boundary layer in the supersonic part of the stream can be thickened, e.g. by injecting a gas into the supersonic part of the stream. The gas can be injected, for example, into the primary cylindrical part of the housing via one or more openings provided in the wall of the housing. Suitably part of the gas from the first outlet is used for this purpose. The effect of such gas-injection is that less condensed particles form in the supersonic part of the stream resulting in larger particles and better agglomeration of the larger particles.

The swirl imparting means can be arranged at the inlet part of the conduit, instead of downstream the inlet part.

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C L A I M S

1. A method of removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components, the method comprising the steps of:

- inducing said stream to flow at supersonic velocity through a conduit so as to decrease the temperature of the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, the conduit being provided with swirl imparting means to impart a swirling motion to the stream of fluid thereby inducing the particles to flow to a radially outer section of a collecting zone in the stream;
- creating a shock wave in the stream so as to decrease the axial velocity of the fluid to subsonic velocity; and
- extracting the particles into an outlet stream from said radially outer section of the collecting zone,

wherein the shock wave is created downstream the swirl imparting means and upstream the collecting zone.

- 20 2. The method of claim 1, wherein the fluid forms a natural gas produced from an earth formation, and said selected temperature is defined by the condensation point of the selected component whereby condensed particles of the selected component are formed.

- 25 3. The method of claim 2, wherein the selected component is one or more of ethane, propane, butane, pentane, hexane, heptane and octane.

- 30 4. The method of any one of claims 1-3, wherein the shock wave is created by inducing the stream of fluid to flow through a diffuser.

5. The method of any one of claims 1-4, wherein said outlet stream is introduced into a gas/liquid separator to separate a gaseous fraction of the outlet stream from a liquid fraction thereof.

5 6. The method of claim 5, wherein said gaseous fraction is mixed with the stream of fluid induced to flow at supersonic velocity through the conduit.

10 7. A device for removing a selected gaseous component from a stream of fluid containing a plurality of gaseous components, comprising:

15 - means for inducing said stream to flow at supersonic velocity through a conduit so as to decrease the temperature of the fluid to below a selected temperature at which one of condensation and solidification of the selected component occurs thereby forming particles of the selected component, the conduit being provided with swirl imparting means to impart a swirling motion to the stream of fluid thereby inducing the particles to flow to a radially outer section of a collecting zone in the

20 stream;

- means for creating a shock wave in the stream so as to decrease the axial velocity of the fluid to subsonic velocity; and

25 - means for extracting the particles into an outlet stream from said radially outer section of the collecting zone, wherein the means for creating the shock wave is arranged to create the shock wave downstream the swirl imparting means and upstream the collecting zone.

8. The device of claim 7, wherein the means for creating the shock wave includes a diffuser.

30 9. The device of claim 8, wherein the means for inducing the stream to flow at a supersonic velocity comprises a Laval-type inlet of the conduit, and wherein the smallest cross-sectional flow area of the diffuser is larger than

the smallest cross-sectional flow area of the Laval-type inlet.

10. The device of claim 8 or 9, wherein said collecting zone is located adjacent the outlet end of the diffuser.

5 11. The device of any one of claims 7-10, further comprising a gas/liquid separator having an inlet in fluid communication with said outlet stream and an outlet for a gaseous fraction of said first outlet stream.

12. The device of claim 11, wherein the outlet for the 10 gaseous fraction is connected to an inlet of the conduit so as to mix the gaseous fraction with the stream of fluid flowing at supersonic velocity through the conduit.

13. The method substantially as described hereinbefore with reference to the drawings.

15 14. The device substantially as described hereinbefore with reference to the drawings.

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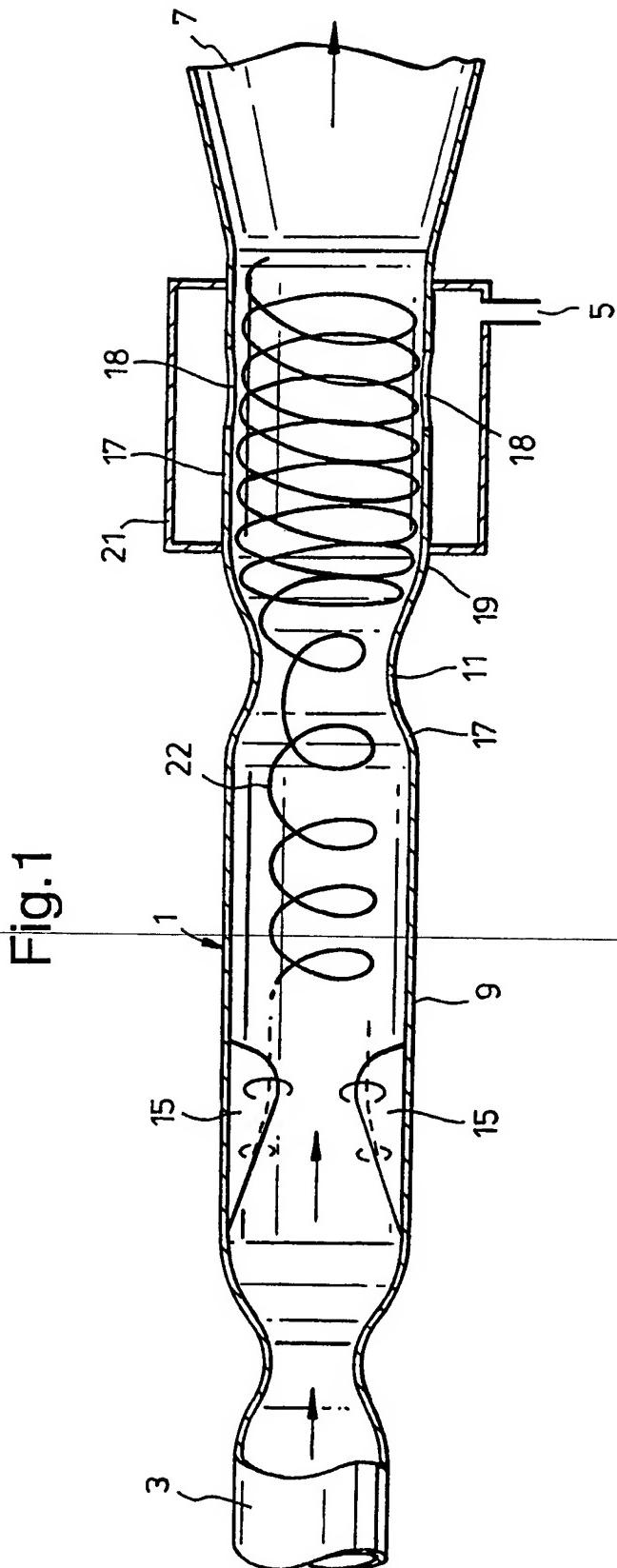


Fig. 1

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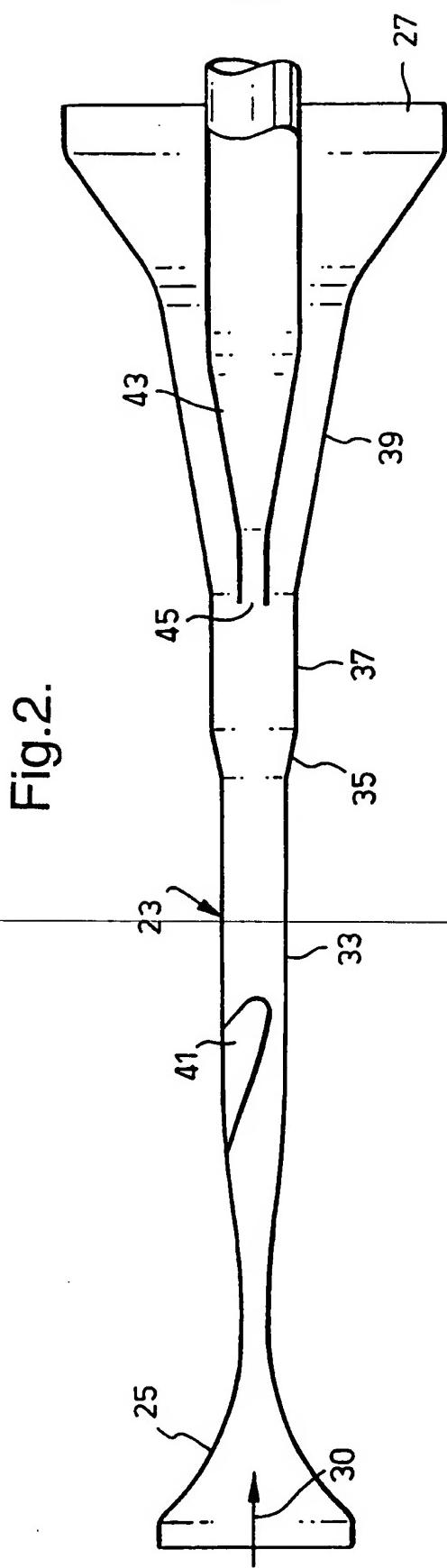


Fig.2.

# INTERNATIONAL SEARCH REPORT

International Application No  
PCT/EP 98/04178

**A. CLASSIFICATION OF SUBJECT MATTER**  
IPC 6 B01D5/00 B01D53/24

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**C. DOCUMENTS CONSIDERED TO BE RELEVANT**

Category *	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
X	PATENT ABSTRACTS OF JAPAN vol. 014, no. 146 (C-0704), 20 March 1990 & JP 02 017921 A (MITSUBISHI HEAVY IND LTD), 22 January 1990 see abstract	1,7,13, 14
Y	US 3 559 373 A (GARRETT ROBERT L) 2 February 1971	4,8-10
A	see abstract; figures 1,10,10A	2,3
A	EP 0 496 128 A (STORK PRODUCT ENGINEERING B.V.) 29 July 1992 see the whole document	1,7
A	EP 0 471 321 A (APRIL DYNAMICS IND LTD) 19 February 1992 see claim 1; figure	1,7
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Patent family members are listed in annex.

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Date of the actual completion of the international search

21 October 1998

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30/10/1998

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International Application No.  
PCT/EP 98/04178

## C.(Continuation) DOCUMENTS CONSIDERED TO BE RELEVANT

Category *	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
A	<p>DATABASE WPI Section Ch, Week 9109 Derwent Publications Ltd., London, GB; Class J08, AN 91-064239 XP002039697 &amp; SU 1 575 047 A (STARKOV V K) , 30 June 1990 see abstract</p> <p>-----</p>	1,7

**INTERNATIONAL SEARCH REPORT**

Information on patent family members

International application No

PCT/EP 96/04178

Patent document cited in search report	Publication date	Patent family member(s)		Publication date
US 3559373	A 02-02-1971	NONE		
EP 0496128	A 29-07-1992	NONE		
EP 0471321	A 19-02-1992	DE 69114783 D		04-01-1996
		DE 69114783 T		25-07-1996

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